Duke University AERO Perseus

2025 IREC 30k-SRAD-Solid Competition

400

Duke AERO

- Began in 2017
- ~ 40 active members
- Major Projects in 24-25:
 - Liquid & Solid Prop.
 - Canards
 - Airbrakes
 - **Vibration-Sensing Fins**
 - Computer Vision Payload
 - **Guided Recovery**



Fig. 1 Team Photo at SAC 2024

Perseus Overview

- First AERO vehicle in the 30k-SRAD category
- Payload ejected @ apogee; uses CV to dodge obstacles
- Third year flying variable-drag airbrakes system
- Second year flying FINSight & SRAD solid motor
- First year flying roll-stabilizing canards system

SRAD Solid Motor



- **APCP** formulation "CP-7" with stepped **BATES** grains
- Hybrid graphite nozzle and aluminum carrier; RTV-sealed

Designations	O-Class, Kn = 220
Burn Time	8.44 s
Total Impulse	~ 35,000 Ns

Fig. 2 SHF O-Class Motor

Airframe

- 135" long & 6" internal diameter
- 120 lbs wet weight; 50.7 lbs dry weight
- 6061 aluminum internal structures; fiberglass tubes
- Forged carbon airbrake petals & canard fins
- G10 fiberglass-core fins; carbon fiber tip-to-tip & epoxy fillets
- All structural components manufactured in-house

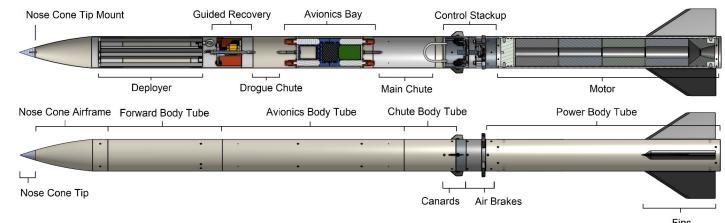


Fig. 3 Perseus CAD, internal & external views

Roll-Stabilizing Canards





Fig. 4a Bevelled Gear Interface Fig. 4b 4-Bar Linkage Interface

- 4-bar linkage + bevel system
- Varies AoA to counteract roll
- Pivot forward CoP by design
- Location informed by CFD
- PI-controlled

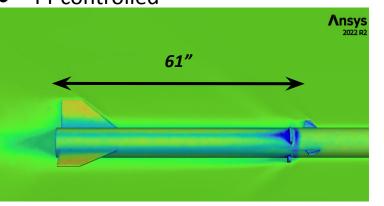


Fig. 5 Dynamic Pressure Contour on Perseus

FINSight Version 2

- 8 accelerometers per fin
- Sense vibrations to inform fin design
- Follows 23-24 *strain gauge* fins
- Accels placement determined by modal simulations
- Wires along external airframe into measurement module

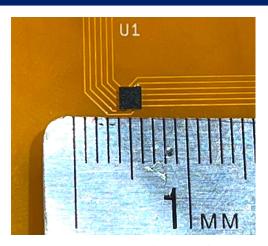
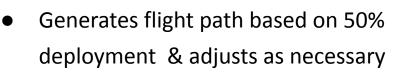


Fig. 6 FINSight Accel

Variable Drag Airbrake System

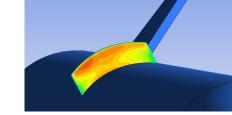


Fig. 8 Dynamic Pressure on



Petal strength validated in flight tests & simulations

Airbrakes



Guided Recovery

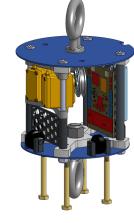
- SRAD cruciform parachute
- PID-controlled dynamic lines



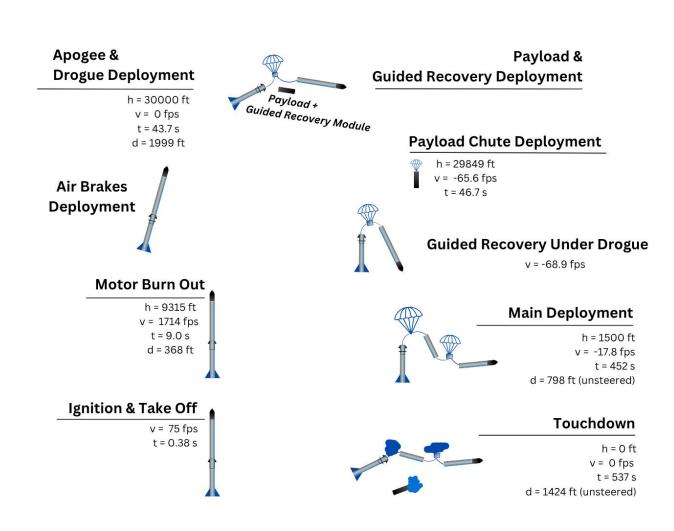


Guided Recovery Module

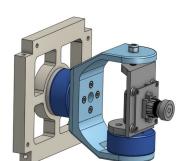
Fig. 10



Concept of Operations (CONOPS)



Minerva (5U CubeSat)



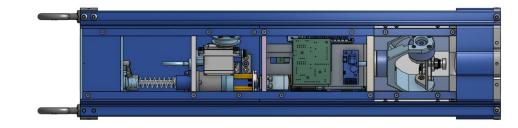


Fig. 11 Gimbal Camera

Fig. 12 Integrated CubeSat and Deployer

- Deployed @ apogee; guided SRAD cruciform parachute
- 2-DOF gimbal camera (s.f. 11) encased in acrylic frame
- Live video & computer vision capabilities

AERO Milestones 24-25

- Critical Design Review (CDR) on January 12
- Ground separation test on April 5
- Static Hot Fire (SHF) on April 5

AERO Sponsors



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⊚ CENTROID[™]

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